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ADDITION TO PROF. R. S. BALL'S PAPER, "NOTE ON A TRANS-FORMATION OF LAGRANGE'S EQUATIONS OF MOTION IN GENERALISED COORDINATES, WHICH IS CONVENIENT IN PHYSICAL ASTRONOMY."

[From the Monthly Notices of the Royal Astronomical Society, vol. XXXVII. (1876—1877), pp. 269—271.]

THE formulæ may be established in a somewhat different way, as follows :--

Consider the masses M_1, M_2, \ldots

Let X_1 , Y_1 , Z_1 be the coordinates (in reference to a fixed origin and axes) of the c.g. of M_1 ;

 x_1, y_1, z_1 the coordinates (in reference to a parallel set of axes through the C.G. of M_1) of an element m_1 of the mass M_1 , and similarly for the masses M_2, \ldots ; the coordinates $(X_1, Y_1, Z_1), (X_2, Y_2, Z_2), \ldots$ all belonging to the same origin and axes;

And let \dot{X}_1 , &c. denote the derived functions $\frac{dX_1}{dt}$, &c.

We have

$$T = S \frac{1}{2} m_1 \left[(\dot{X}_1 + \dot{x}_1)^2 + (\dot{Y}_1 + \dot{y}_1)^2 + (\dot{Z}_1 + \dot{z}_1)^2 \right] + S \frac{1}{2} m_2 \left[(\dot{X}_2 + \dot{x}_2)^2 + (\dot{Y}_2 + \dot{y}_2)^2 + (\dot{Z}_2 + \dot{z}_2)^2 \right]$$

or since $Sm_1x_1 = 0$, &c., and therefore also $Sm_1\dot{x}_1 = 0$, &c., this is

$$T = \frac{1}{2}M_1 \left(\dot{X}_1^2 + \dot{Y}_1^2 + \dot{Z}_1^2 \right) + S \frac{1}{2}m_1 \left(\dot{x}_1^2 + \dot{y}_1^2 + \dot{z}_1^2 \right) \\ + \frac{1}{2}M_2 \left(\dot{X}_2^2 + \dot{Y}_2^2 + \dot{Z}_2^2 \right) + S \frac{1}{2}m_2 \left(\dot{x}_2^2 + \dot{y}_2^2 + \dot{z}_2^2 \right)$$

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Write u, v, w for the coordinates of the C.G. of the whole system: then

$$\begin{split} M_1 X_1 + M_2 X_2 + \ldots &= (M_1 + M_2 \ldots) u, \\ M_1 Y_1 + M_2 Y_2 + \ldots &= (M_1 + M_2 \ldots) v, \\ M_1 Z_1 + M_2 Z_2 + \ldots &= (M_1 + M_2 \ldots) w; \\ M_1 \dot{X}_1 + M_2 \dot{X}_2 + \ldots &= (M_1 + M_2 \ldots) \dot{u}, \\ M_1 \dot{Y}_1 + M_2 \dot{Y}_2 + \ldots &= (M_1 + M_2 \ldots) \dot{v}, \\ M_1 \dot{Z}_1 + M_2 \dot{Z}_2 + \ldots &= (M_1 + M_2 \ldots) \dot{v}; \end{split}$$

and thence

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T

$$\begin{split} &= \frac{1}{M_1 + M_2 + \dots} (\dot{u}^2 + \dot{v}^2 + w^2) \\ &= \frac{1}{M_1 + M_2 \dots} \left\{ M_1 M_2 \left[(\dot{X}_1 - \dot{X}_2)^2 + (\dot{Y}_1 - \dot{Y}_2)^2 + (\dot{Z}_1 - \dot{Z}_2)^2 \right] \right\} \\ &: \\ &+ S \frac{1}{2} m_1 (\dot{x}_1^2 + \dot{y}_1^2 + \dot{z}_1^2) \\ &+ S \frac{1}{2} m_2 (\dot{x}_2^2 + \dot{y}_2^2 + \dot{z}_2^2) \\ &: \\ &: \end{split}$$

or, representing the function on the right-hand side by T', this is

$$T = \frac{1}{2} \left(M_1 + M_2 + \dots \right) \left(\dot{u}_2 + \dot{v}_2 + \dot{w}_2 \right) + T' \dots, = T_0 + T'.$$

Suppose the positions are determined by means of the 6n coordinates ((q)); the equations of motion are each of them of the form

$$\frac{d}{dt} \cdot \frac{dT_{\circ}}{d\dot{q}} - \frac{dT_{\circ}}{dq} + \frac{d}{dt} \cdot \frac{dT'}{d\dot{q}} - \frac{dT'}{dq} = -\frac{dV}{dq}.$$

But these admit of further reduction; the part in T_0 depends upon three terms, such as

$$rac{d}{dt}\left(\dot{u}rac{d\dot{u}}{d\dot{q}}
ight)-\dot{u}rac{d\dot{u}}{dq},\ =rac{d\dot{u}}{dt}rac{d\dot{u}}{d\dot{q}}+\dot{u}\left(rac{d}{dt}rac{d\dot{u}}{d\dot{q}}-rac{d\dot{u}}{dq}
ight).$$

But we have u a function of ((q)), and thence

$$\frac{d\dot{u}}{d\dot{q}} = \frac{du}{dq}, \text{ or } \frac{d}{dt}\frac{d\dot{u}}{d\dot{q}} - \frac{d\dot{u}}{dq}, = \frac{d}{dt}\frac{du}{dq} - \frac{d\dot{u}}{dq}, = 0,$$

or the term is simply

$$=\frac{d\dot{u}}{dt}\,\frac{d\dot{u}}{dq}.$$

The equation thus becomes

$$(M_1 + M_2 \dots) \left(\frac{d\dot{u}}{dt} \frac{d\dot{u}}{d\dot{q}} + \frac{d\dot{v}}{dt} \frac{d\dot{v}}{d\dot{q}} + \frac{d\dot{w}}{dt} \frac{d\dot{w}}{d\dot{q}} \right) + \frac{d}{dt} \frac{dT'}{d\dot{q}} - \frac{dT'}{dq} = -\frac{dV}{dq}$$

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Suppose now that T', V are functions of 6n-3 out of the 6n coordinates ((q)), and of the differential coefficients \dot{q} of the same 6n-3 coordinates, but are independent of the remaining three coordinates and of their differential coefficients; then, first, if q denotes any one of the three coordinates, the equation becomes

$$\frac{d\dot{u}}{dt}\frac{d\dot{u}}{d\dot{q}} + \frac{d\dot{v}}{dt}\frac{d\dot{v}}{d\dot{q}} + \frac{d\dot{w}}{dt}\frac{d\dot{w}}{d\dot{q}} = 0;$$

or, better,

$$\frac{d\dot{u}}{dt}\frac{du}{dq} + \frac{d\dot{v}}{dt}\frac{dv}{dq} + \frac{d\dot{w}}{dt}\frac{dw}{dq} = 0;$$

and the three equations of this form give

$$\frac{d\dot{u}}{dt} = 0, \quad \frac{d\dot{v}}{dt} = 0, \quad \frac{d\dot{w}}{dt} = 0,$$

viz. these are the equations for the conservation of the motion of the centre of gravity.

And this being so, then, if q now denotes any one of the 6n-3 coordinates, each of the remaining equations assumes the form

$$\frac{d}{dt} \cdot \frac{dT'}{d\dot{q}} - \frac{dT'}{dq} = -\frac{dV}{dq},$$

viz. we have thus 6n - 3 equations for the relative motion of the bodies of the system.

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